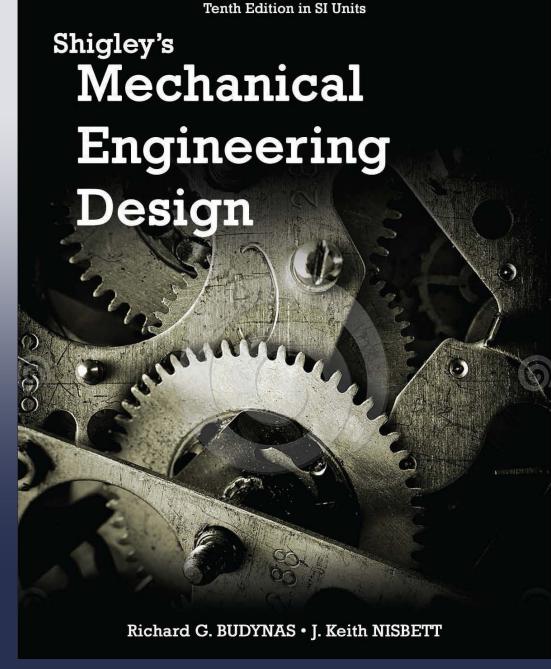


Lecture Slides

Chapter 6

Fatigue Failure Resulting from Variable Loading



Completely Reversing Simple Loading

1 Determine S'_e either from test data or

p. 274
$$S'_{e} = \begin{cases} 0.5S_{ut} & S_{ut} \leq 200 \text{ kpsi } (1400 \text{ MPa}) \\ 100 \text{ kpsi} & S_{ut} > 200 \text{ kpsi} \\ 700 \text{ MPa} & S_{ut} > 1400 \text{ MPa} \end{cases}$$
 (6–8)

2 Modify S'_e to determine S_e .

p. 279
$$S_e = k_a k_b k_c k_d k_e k_f S'_e$$
 (6–18)
$$k_a = a S^b_{ut}$$
 (6–19)

Table 6-2

Parameters for Marin Surface Modification Factor, Eq. (6–19)

Surface	Factor a		Exponent
Finish	S _{ut} , kpsi	S _{ut} , MPa	ь
Ground	1.34	1.58	-0.085
Machined or cold-drawn	2.70	4.51	-0.265
Hot-rolled	14.4	57.7	-0.718
As-forged	39.9	272.	-0.995

Rotating shaft. For bending or torsion,

p. 280
$$k_b = \begin{cases} (d/0.3)^{-0.107} = 0.879d^{-0.107} & 0.11 \le d \le 2 \text{ in} \\ 0.91d^{-0.157} & 2 < d \le 10 \text{ in} \\ (d/7.62)^{-0.107} = 1.24d^{-0.107} & 2.79 \le d \le 51 \text{ mm} \\ 1.51d^{-0.157} & 51 < 254 \text{ mm} \end{cases}$$
(6–20)

For axial,

$$k_b = 1$$
 (6–21)

Nonrotating member. Use Table 6–3, p. 282, for d_e and substitute into Eq. (6–20) for d.

p. 282
$$k_c = \begin{cases} 1 & \text{bending} \\ 0.85 & \text{axial} \\ 0.59 & \text{torsion} \end{cases}$$
 (6–26)

p. 283 Use Table 6–4 for k_d , or

$$k_d = 0.975 + 0.432(10^{-3})T_F - 0.115(10^{-5})T_F^2 + 0.104(10^{-8})T_F^3 - 0.595(10^{-12})T_F^4$$
(6-27)

pp. 284–285, k_e

Table 6-5

Reliability Factors k_e Corresponding to
8 Percent Standard
Deviation of the
Endurance Limit

Reliability, %	Transformation Variate z_a	Reliability Factor k_e
50	0	1.000
90	1.288	0.897
95	1.645	0.868
99	2.326	0.814
99.9	3.091	0.753
99.99	3.719	0.702
99.999	4.265	0.659
99.9999	4.753	0.620

pp. 285–286,
$$k_f$$

3 Determine fatigue stress-concentration factor, K_f or K_{fs} . First, find K_t or K_{ts} from Table A–15.

p. 287
$$K_f = 1 + q(K_t - 1)$$
 or $K_{fs} = 1 + q(K_{ts} - 1)$ (6–32)

Obtain q from either Fig. 6–20 or 6–21, pp. 287–288.

Alternatively, for reversed bending or axial loads,

p. 288
$$K_f = 1 + \frac{K_t - 1}{1 + \sqrt{a/r}}$$
 (6–33)

For S_{ut} in kpsi,

$$\sqrt{a} = 0.245799 - 0.307794(10^{-2})S_{ut}$$

+0.150874(10⁻⁴) $S_{ut}^2 - 0.266978(10^{-7})S_{ut}^3$ (6–35)

For torsion for low-alloy steels, increase S_{ut} by 20 kpsi and apply to Eq. (6–35).

- 4 Apply K_f or K_{fs} by either dividing S_e by it or multiplying it with the purely reversing stress not both.
- Determine fatigue life constants a and b. If $S_{ut} \ge 70$ kpsi, determine f from Fig. 6–18, p. 277. If $S_{ut} < 70$ kpsi, let f = 0.9.

p. 277
$$a = (f S_{ut})^2 / S_e$$
 (6–14)

$$b = -[\log(f S_{ut}/S_e)]/3 \tag{6-15}$$

6 Determine fatigue strength S_f at N cycles, or, N cycles to failure at a reversing stress σ_a

(*Note*: this only applies to purely reversing stresses where $\sigma_m = 0$).

p. 277
$$S_f = aN^b$$
 (6–13)

$$N = (\sigma_a/a)^{1/b} {(6-16)}$$

Fluctuating Simple Loading

For S_e , K_f or K_{fs} , see previous subsection.

1 Calculate σ_m and σ_a . Apply K_f to both stresses.

p. 293
$$\sigma_m = (\sigma_{\text{max}} + \sigma_{\text{min}})/2$$
 $\sigma_a = |\sigma_{\text{max}} - \sigma_{\text{min}}|/2$ (6–36)

2 Apply to a fatigue failure criterion, p. 298

$$\sigma_m \geq 0$$

Soderburg
$$\sigma_a/S_e + \sigma_m/S_y = 1/n$$
 (6–45)
mod-Goodman $\sigma_a/S_e + \sigma_m/S_{ut} = 1/n$ (6–46)
Gerber $n\sigma_a/S_e + (n\sigma_m/S_{ut})^2 = 1$ (6–47)
ASME-elliptic $(\sigma_a/S_e)^2 + (\sigma_m/S_{ut})^2 = 1/n^2$ (6–48)

$$\sigma_m < 0$$

p. 297
$$\sigma_a = S_e/n$$

Torsion. Use the same equations as apply for $\sigma_m \ge 0$, except replace σ_m and σ_a with τ_m and τ_a , use $k_c = 0.59$ for S_e , replace S_{ut} with $S_{su} = 0.67S_{ut}$ [Eq. (6–54), p. 309], and replace S_v with $S_{sv} = 0.577S_v$ [Eq. (5–21), p. 217]

3 Check for localized yielding.

p. 298
$$\sigma_a + \sigma_m = S_y/n$$
 (6–49) or, for torsion,
$$\tau_a + \tau_m = 0.577S_y/n$$

4 For finite-life fatigue strength (see Ex. 6–12, pp. 305–306),

mod-Goodman
$$S_f = \frac{\sigma_a}{1 - (\sigma_m/S_{ut})}$$
 Gerber
$$S_f = \frac{\sigma_a}{1 - (\sigma_m/S_{ut})^2}$$

If determining the finite life N with a factor of safety n, substitute S_f/n for σ_a in Eq. (6–16). That is,

$$N = \left(\frac{S_f/n}{a}\right)^{1/b}$$

Combination of Loading Modes

See previous subsections for earlier definitions.

1 Calculate von Mises stresses for alternating and midrange stress states, σ'_a and σ'_m . When determining S_e , do not use k_c nor divide by K_f or K_{fs} . Apply K_f and/or K_{fs} directly to each specific alternating and midrange stress. If axial stress is present divide the alternating axial stress by $k_c = 0.85$. For the special case of combined bending, torsional shear, and axial stresses

p. 310

$$\sigma_a' = \left\{ \left[(K_f)_{bending}(\sigma_a)_{bending} + (K_f)_{axial} \frac{(\sigma_a)_{axial}}{0.85} \right]^2 + 3 \left[(K_{fs})_{torsion}(\tau_a)_{torsion} \right]^2 \right\}^{1/2}$$

$$(6-55)$$

$$\sigma'_{m} = \left\{ \left[(K_{f})_{bending}(\sigma_{m})_{bending} + (K_{f})_{axial}(\sigma_{m})_{axial} \right]^{2} + 3 \left[(K_{fs})_{torsion}(\tau_{m})_{torsion} \right]^{2} \right\}^{1/2}$$

$$(6-56)$$

- 2 Apply stresses to fatigue criterion [see Eq. (6–45) to (6–48), p. 338 in previous subsection].
- 3 Conservative check for localized yielding using von Mises stresses.

p. 298
$$\sigma'_a + \sigma'_m = S_y/n$$
 (6–49)